

6.1-4 - Laminate Constitutive Equations

$$(6.1-10) \quad \begin{Bmatrix} N_x \\ N_y \\ N_{xy} \end{Bmatrix} = [A] \begin{Bmatrix} \varepsilon_x^0 \\ \varepsilon_y^0 \\ \gamma_{xy}^0 \end{Bmatrix} + [B] \begin{Bmatrix} \varepsilon_x^{(1)} \\ \varepsilon_y^{(1)} \\ \gamma_{xy}^{(1)} \end{Bmatrix} \quad (6.1-28)$$

$$\begin{Bmatrix} M_x \\ M_y \\ M_{xy} \end{Bmatrix} = [B] \begin{Bmatrix} \varepsilon_x^0 \\ \varepsilon_y^0 \\ \gamma_{xy}^0 \end{Bmatrix} + [D] \begin{Bmatrix} \varepsilon_x^{(1)} \\ \varepsilon_y^{(1)} \\ \gamma_{xy}^{(1)} \end{Bmatrix} \quad (6.1-29)$$

$$\begin{Bmatrix} \bar{N} \\ \bar{M} \end{Bmatrix} = \begin{bmatrix} A & | & B \\ \hline B & | & D \end{bmatrix} \begin{Bmatrix} \varepsilon^0 \\ \varepsilon^{(1)} \end{Bmatrix} \quad (6.1-30)$$

$$\begin{Bmatrix} \bar{N} \\ \bar{M} \end{Bmatrix} = \begin{Bmatrix} N \\ M \end{Bmatrix} + \begin{Bmatrix} N^T \\ M^T \end{Bmatrix} + \begin{Bmatrix} N^P \\ M^P \end{Bmatrix} \quad (6.1-31)$$

$$\{N^T\} = \sum_{k=1}^N \int_{z_k}^{z_{k-1}} [\bar{Q}]_k \{\bar{\alpha}\}_k \Delta T dz \quad (6.1-32)$$

$$\{M^T\} = \sum_{k=1}^N \int_{z_k}^{z_{k-1}} [\bar{Q}]_k \{\bar{\alpha}\}_k \Delta T z dz$$

$$\{N^P\} = \sum_{k=1}^N \int_{z_k}^{z_{k-1}} [\bar{e}]_k \{e\}_k dz$$

(6.1-33)

$$\{M^P\} = \sum_{k=1}^N \int_{z_k}^{z_{k-1}} [\bar{e}]_k \{e\}_k dz$$

$\{N^P\}$  and  $\{M^P\}$  are the piezoelectric resultants.

## 6.1-5. Equations of motions in Terms of Displacements

we had:

$$\{N\} = [A] \left\{ \begin{array}{l} \frac{\partial u_0}{\partial x} + \frac{1}{2} \left( \frac{\partial w_0}{\partial x} \right)^2 \\ \frac{\partial v_0}{\partial y} + \frac{1}{2} \left( \frac{\partial w_0}{\partial y} \right)^2 \\ \frac{\partial u_0}{\partial y} + \frac{\partial v_0}{\partial x} + \frac{\partial w_0}{\partial x} \frac{\partial w_0}{\partial y} \end{array} \right\} - [B] \left\{ \begin{array}{l} \frac{\partial^2 w_0}{\partial x^2} \\ \frac{\partial^2 w_0}{\partial y^2} \\ 2 \frac{\partial^2 w_0}{\partial x \partial y} \end{array} \right\} \quad (6.1-34)$$

$$\{M\} = [B] \left\{ \begin{array}{l} \cdot \\ \cdot \\ \cdot \end{array} \right\} - [D] \left\{ \begin{array}{l} \cdot \\ \cdot \\ \cdot \end{array} \right\} \quad (6.1-35)$$

substituting these equations in the governing equations (6.1-21) we have:

$$\begin{aligned}
& A_{11} \left( \frac{\partial^2 u_0}{\partial x^2} + \frac{\partial w_0}{\partial x} \frac{\partial^2 w_0}{\partial x^2} \right) + A_{12} \left( \frac{\partial^2 v_0}{\partial x \partial y} + \frac{\partial w_0}{\partial y} \frac{\partial^2 w_0}{\partial x \partial y} \right) \\
& + A_{16} \left( \frac{\partial^2 u_0}{\partial x \partial y} + \frac{\partial^2 v_0}{\partial x^2} + \frac{\partial^2 w_0}{\partial x^2} \frac{\partial w_0}{\partial y} + \frac{\partial w_0}{\partial x} \frac{\partial^2 w_0}{\partial x \partial y} \right) \\
& - B_{11} \frac{\partial^3 w_0}{\partial x^3} - B_{12} \frac{\partial^3 w_0}{\partial x \partial y^2} - 2B_{16} \frac{\partial^3 w_0}{\partial x^2 \partial y} \\
& + A_{16} \left( \frac{\partial^2 u_0}{\partial x \partial y} + \frac{\partial w_0}{\partial x} \frac{\partial^2 w_0}{\partial x \partial y} \right) + A_{26} \left( \frac{\partial^2 v_0}{\partial y^2} + \frac{\partial w_0}{\partial y} \frac{\partial^2 w_0}{\partial y^2} \right) \\
& + A_{66} \left( \frac{\partial^2 u_0}{\partial y^2} + \frac{\partial^2 v_0}{\partial x \partial y} + \frac{\partial^2 w_0}{\partial x \partial y} \frac{\partial w_0}{\partial y} + \frac{\partial w_0}{\partial x} \frac{\partial^2 w_0}{\partial y^2} \right) \\
& - B_{16} \frac{\partial^3 w_0}{\partial x^2 \partial y} - B_{26} \frac{\partial^3 w_0}{\partial y^3} - 2B_{66} \frac{\partial^3 w_0}{\partial x \partial y^2} \\
& - \left( \frac{\partial N_{xx}^T}{\partial x} + \frac{\partial N_{xy}^T}{\partial y} \right) = I_0 \frac{\partial^2 u_0}{\partial t^2} - I_1 \frac{\partial^3 w_0}{\partial x \partial t^2}
\end{aligned}$$

(6.1-36a)

$$\begin{aligned}
& A_{16} \left( \frac{\partial^2 u_0}{\partial x^2} + \frac{\partial w_0}{\partial x} \frac{\partial^2 w_0}{\partial x^2} \right) + A_{26} \left( \frac{\partial^2 v_0}{\partial x \partial y} + \frac{\partial w_0}{\partial y} \frac{\partial^2 w_0}{\partial x \partial y} \right) \\
& + A_{66} \left( \frac{\partial^2 u_0}{\partial x \partial y} + \frac{\partial^2 v_0}{\partial x^2} + \frac{\partial^2 w_0}{\partial x^2} \frac{\partial w_0}{\partial y} + \frac{\partial w_0}{\partial x} \frac{\partial^2 w_0}{\partial x \partial y} \right) \\
& - B_{16} \frac{\partial^3 w_0}{\partial x^3} - B_{26} \frac{\partial^3 w_0}{\partial x \partial y^2} - 2B_{66} \frac{\partial^3 w_0}{\partial x^2 \partial y} \\
& + A_{12} \left( \frac{\partial^2 u_0}{\partial x \partial y} + \frac{\partial w_0}{\partial x} \frac{\partial^2 w_0}{\partial x \partial y} \right) + A_{22} \left( \frac{\partial^2 v_0}{\partial y^2} + \frac{\partial w_0}{\partial y} \frac{\partial^2 w_0}{\partial y^2} \right) \\
& + A_{26} \left( \frac{\partial^2 u_0}{\partial y^2} + \frac{\partial^2 v_0}{\partial x \partial y} + \frac{\partial^2 w_0}{\partial x \partial y} \frac{\partial w_0}{\partial y} + \frac{\partial w_0}{\partial x} \frac{\partial^2 w_0}{\partial y^2} \right) \\
& - B_{12} \frac{\partial^3 w_0}{\partial x^2 \partial y} - B_{22} \frac{\partial^3 w_0}{\partial y^3} - 2B_{26} \frac{\partial^3 w_0}{\partial x \partial y^2} \\
& - \left( \frac{\partial N_{xy}^T}{\partial x} + \frac{\partial N_{yy}^T}{\partial y} \right) = I_0 \frac{\partial^2 v_0}{\partial t^2} - I_1 \frac{\partial^3 w_0}{\partial y \partial t^2}
\end{aligned}$$

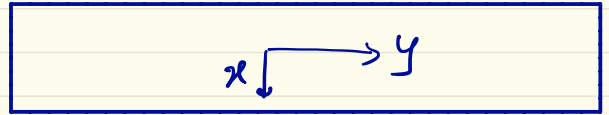
(6.1-36b)

$$\begin{aligned}
& B_{11} \left( \frac{\partial^3 v_0}{\partial x^3} + \frac{\partial^2 w_0}{\partial x^2} \frac{\partial^2 w_0}{\partial x^2} + \frac{\partial w_0}{\partial x} \frac{\partial^3 w_0}{\partial x^3} \right) + B_{12} \left( \frac{\partial^3 v_0}{\partial x^2 \partial y} + \frac{\partial^2 w_0}{\partial x \partial y} \frac{\partial^2 w_0}{\partial x \partial y} \right. \\
& \left. + \frac{\partial w_0}{\partial y} \frac{\partial^3 w_0}{\partial x^2 \partial y} \right) + B_{16} \left( \frac{\partial^3 v_0}{\partial x^2 \partial y} + \frac{\partial^3 v_0}{\partial x^3} + \frac{\partial^3 w_0}{\partial x^3} \frac{\partial w_0}{\partial y} + 2 \frac{\partial^2 w_0}{\partial x^2} \frac{\partial^2 w_0}{\partial x \partial y} \right. \\
& \left. + \frac{\partial w_0}{\partial x} \frac{\partial^3 w_0}{\partial x^2 \partial y} \right) - D_{11} \frac{\partial^4 w_0}{\partial x^4} - D_{12} \frac{\partial^4 w_0}{\partial x^2 \partial y^2} - 2D_{16} \frac{\partial^4 w_0}{\partial x^3 \partial y} \\
& + 2B_{16} \left( \frac{\partial^3 v_0}{\partial x^2 \partial y} + \frac{\partial^2 w_0}{\partial x^2} \frac{\partial^2 w_0}{\partial x \partial y} + \frac{\partial w_0}{\partial x} \frac{\partial^3 w_0}{\partial x^2 \partial y} \right) + 2B_{26} \left( \frac{\partial^3 v_0}{\partial x \partial y^2} + \frac{\partial^2 w_0}{\partial x \partial y} \frac{\partial^2 w_0}{\partial y^2} \right. \\
& \left. + \frac{\partial w_0}{\partial y} \frac{\partial^3 w_0}{\partial x \partial y^2} \right) + 2B_{66} \left( \frac{\partial^3 v_0}{\partial x \partial y^2} + \frac{\partial^3 v_0}{\partial x^2 \partial y} + \frac{\partial^3 w_0}{\partial x^2 \partial y} \frac{\partial w_0}{\partial y} + \frac{\partial^2 w_0}{\partial x \partial y} \frac{\partial^2 w_0}{\partial x \partial y} \right. \\
& \left. + \frac{\partial^2 w_0}{\partial x^2} \frac{\partial^2 w_0}{\partial y^2} + \frac{\partial w_0}{\partial x} \frac{\partial^3 w_0}{\partial x \partial y^2} \right) - 2D_{16} \frac{\partial^4 w_0}{\partial x^3 \partial y} - 2D_{26} \frac{\partial^4 w_0}{\partial x \partial y^3} \\
& - 4D_{66} \frac{\partial^4 w_0}{\partial x^2 \partial y^2} + B_{12} \left( \frac{\partial^3 v_0}{\partial x \partial y^2} + \frac{\partial^2 w_0}{\partial x \partial y} \frac{\partial^2 w_0}{\partial x \partial y} + \frac{\partial w_0}{\partial x} \frac{\partial^3 w_0}{\partial x \partial y^2} \right) \\
& + B_{22} \left( \frac{\partial^3 v_0}{\partial y^3} + \frac{\partial^2 w_0}{\partial y^2} \frac{\partial^2 w_0}{\partial y^2} + \frac{\partial w_0}{\partial y} \frac{\partial^3 w_0}{\partial y^3} \right) + B_{26} \left( \frac{\partial^3 v_0}{\partial y^3} + \frac{\partial^3 v_0}{\partial x \partial y^2} \right. \\
& \left. + \frac{\partial^3 w_0}{\partial x \partial y^2} \frac{\partial w_0}{\partial y} + 2 \frac{\partial^2 w_0}{\partial x \partial y} \frac{\partial^2 w_0}{\partial y^2} + \frac{\partial w_0}{\partial x} \frac{\partial^3 w_0}{\partial y^3} \right) \\
& - D_{12} \frac{\partial^4 w_0}{\partial x^2 \partial y^2} - D_{22} \frac{\partial^4 w_0}{\partial y^4} - 2D_{26} \frac{\partial^4 w_0}{\partial x \partial y^3} + \mathcal{N}(w_0) + q \\
& - \left( \frac{\partial^2 M_{xx}^T}{\partial x^2} + 2 \frac{\partial^2 M_{xy}^T}{\partial y \partial x} + \frac{\partial^2 M_{yy}^T}{\partial y^2} \right) \\
& = I_0 \frac{\partial^2 w_0}{\partial t^2} - I_2 \frac{\partial^2}{\partial t^2} \left( \frac{\partial^2 w_0}{\partial x^2} + \frac{\partial^2 w_0}{\partial y^2} \right) + I_1 \frac{\partial^2}{\partial t^2} \left( \frac{\partial w_0}{\partial x} + \frac{\partial w_0}{\partial y} \right)
\end{aligned}$$

(6.1-36c)

## Example: (Cylindrical Bending)

In a plate when we have infinite dimension along the  $y$ -axis and finite dimension along the  $x$ -axis and subjected to a transverse load  $q(x)$



$$A_{11} \left( \frac{\partial^2 u_0}{\partial x^2} + \frac{\partial w_0}{\partial x} \frac{\partial^2 w_0}{\partial x^2} \right) + A_{16} \frac{\partial^2 v_0}{\partial x^2} - B_{11} \frac{\partial^3 w_0}{\partial x^3} - \frac{\partial N_{xx}^T}{\partial x} = I_0 \frac{\partial^2 u_0}{\partial t^2} - I_1 \frac{\partial^3 w_0}{\partial x \partial t^2} \quad \left( \frac{\partial}{\partial y} = 0 \right)$$

$$A_{16} \left( \frac{\partial^2 u_0}{\partial x^2} + \frac{\partial w_0}{\partial x} \frac{\partial^2 w_0}{\partial x^2} \right) + A_{66} \frac{\partial^2 v_0}{\partial x^2} - B_{16} \frac{\partial^3 w_0}{\partial x^3} - \frac{\partial N_{xy}^T}{\partial x} = I_0 \frac{\partial^2 v_0}{\partial t^2}$$

$$B_{11} \left( \frac{\partial^3 u_0}{\partial x^3} + \frac{\partial^2 w_0}{\partial x^2} \frac{\partial^2 w_0}{\partial x^2} + \frac{\partial w_0}{\partial x} \frac{\partial^3 w_0}{\partial x^3} \right) + B_{16} \frac{\partial^3 v_0}{\partial x^3} - D_{11} \frac{\partial^4 w_0}{\partial x^4} - \frac{\partial^2 M_{xx}^T}{\partial x^2}$$

$$+ \frac{\partial}{\partial x} \left( N_{xx} \frac{\partial w_0}{\partial x} \right) + q = I_0 \frac{\partial^2 w_0}{\partial t^2} - I_2 \frac{\partial^4 w_0}{\partial x^2 \partial t^2} + I_1 \frac{\partial^3 u_0}{\partial x \partial t^2}$$

(6.1-37)

## 6.2. The First-Order Laminated Plate Theory (FSPT)

first-order shear deformation laminated plate theory.

### 6.2.1. Displacements and Strains

$$u(x, y, z, t) = u_0(x, y, t) + z \phi_x(x, y, t) \quad (6.2-1)$$

$$v(x, y, z, t) = v_0(x, y, t) + z \phi_y(x, y, t)$$

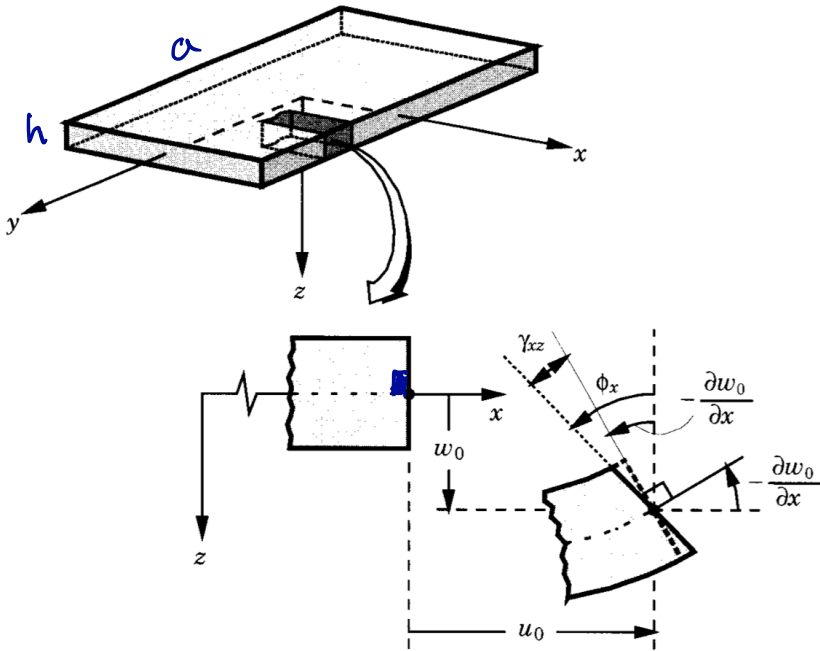
$$w(x, y, z, t) = w_0(x, y, t)$$

where  $(u_0, v_0, w_0, \phi_x, \phi_y)$  are unknown functions to be determined. As before,  $(u_0, v_0, w_0)$  denote the displacements of a point on the plane  $z=0$ . Note that

$$\phi_x = \frac{\partial u}{\partial z}, \quad \phi_y = \frac{\partial v}{\partial z} \quad (6.2-2)$$



which indicate that  $\phi_x$  and  $\phi_y$  are the rotations of a transverse normal about the  $y$ - and  $x$ -axes.



if  $\frac{a}{h} \gg 50$  then

$$\phi_x = -\frac{\partial w_0}{\partial x}, \quad \phi_y = -\frac{\partial w_0}{\partial y}$$

nonlinear strains:

$$\epsilon_x = \frac{\partial u_0}{\partial x} + \frac{1}{2} \left( \frac{\partial w_0}{\partial x} \right)^2 + z \frac{\partial \phi_x}{\partial x}$$

linear through the thickness

$$\epsilon_y = \frac{\partial v_0}{\partial y} + \frac{1}{2} \left( \frac{\partial w_0}{\partial y} \right)^2 + z \frac{\partial \phi_y}{\partial y}$$

linear through the thickness

$$\gamma_{xy} = \left( \frac{\partial u_0}{\partial y} + \frac{\partial v_0}{\partial x} + \frac{\partial w_0}{\partial x} \frac{\partial w_0}{\partial y} \right) + z \left( \frac{\partial \phi_x}{\partial y} + \frac{\partial \phi_y}{\partial x} \right)$$

linear through the thickness

$$\gamma_{xz} = \frac{\partial w_0}{\partial x} + \phi_x$$

(6.2-4)

$$\gamma_{yz} = \frac{\partial w_0}{\partial y} + \phi_y$$

constant through the thickness (approximation)

$$\epsilon_z = 0$$

(6.2-5)

$$\begin{Bmatrix} \epsilon_x \\ \epsilon_y \\ \epsilon_z \\ \gamma_{yz} \\ \gamma_{xz} \\ \gamma_{xy} \end{Bmatrix} = \begin{Bmatrix} \epsilon_x^0 \\ \epsilon_y^0 \\ \epsilon_z^0 \\ \gamma_{yz}^0 \\ \gamma_{xz}^0 \\ \gamma_{xy}^0 \end{Bmatrix} + z \begin{Bmatrix} \epsilon_x^{(1)} \\ \epsilon_y^{(1)} \\ \epsilon_z^{(1)} \\ \gamma_{yz}^{(1)} \\ \gamma_{xz}^{(1)} \\ \gamma_{xy}^{(1)} \end{Bmatrix} = \begin{Bmatrix} \frac{\partial u_0}{\partial x} + \frac{1}{2} \left( \frac{\partial w_0}{\partial x} \right)^2 \\ \frac{\partial v_0}{\partial y} + \frac{1}{2} \left( \frac{\partial w_0}{\partial y} \right)^2 \\ \frac{\partial w_0}{\partial x} + \phi_x \\ \frac{\partial w_0}{\partial y} + \phi_y \\ \frac{\partial u_0}{\partial y} + \frac{\partial v_0}{\partial x} + \frac{\partial w_0}{\partial x} \frac{\partial w_0}{\partial y} \end{Bmatrix} + z \begin{Bmatrix} \frac{\partial \phi_x}{\partial x} \\ \frac{\partial \phi_y}{\partial y} \\ 0 \\ 0 \\ \frac{\partial \phi_x}{\partial y} + \frac{\partial \phi_y}{\partial x} \end{Bmatrix}$$

## 6.2.2 Equations of Motion

$$\delta \mathcal{L} = \int_0^T (\delta U + \delta V - \delta K) dt \quad (6.2-6)$$

$$\delta U = \int_{\Omega_0} \left\{ \int_{-h/2}^{h/2} [\sigma_x (\delta \epsilon_x^\circ + z \delta \epsilon_x^{(1)}) + \sigma_y (\delta \epsilon_y^\circ + z \delta \epsilon_y^{(1)}) + \sigma_{xy} (\delta \gamma_{xy}^\circ + z \delta \gamma_{xy}^{(1)}) + \sigma_{xz} \delta \gamma_{xz}^\circ + \sigma_{yz} \delta \gamma_{yz}^\circ] dz \right\} dx dy$$

$$\delta V = - \int_{\Omega_0} [(q_b, q_t) \delta w_0] dx dy - \int_{\Gamma_0} \int_{-h/2}^{h/2} [\hat{\sigma}_n (\delta u_n + z \delta \phi_n) + \hat{\sigma}_{ns} (\delta u_s + z \delta \phi_s) + \hat{\sigma}_{nz} \delta w_0] ds dz$$

$$\delta K = \int_{\Omega_0} \int_{-h/2}^{h/2} \rho_0 [(u_0 + z \dot{\phi}_x) (\delta u_0 + z \delta \dot{\phi}_x) + (v_0 + z \dot{\phi}_y) (\delta v_0 + z \delta \dot{\phi}_y) + w_0 \delta \dot{w}_0] dz dx dy$$

$$(6.2-7)$$

Substituting (6.2-7) into (6.2.6) and integrating through the thickness we obtain:

$$0 = \int_0^T \left\{ \int_{\Omega_0} \left[ N_{xx} \delta \varepsilon_{xx}^{(0)} + M_{xx} \delta \varepsilon_{xx}^{(1)} + N_{yy} \delta \varepsilon_{yy}^{(0)} + M_{yy} \delta \varepsilon_{yy}^{(1)} + N_{xy} \delta \gamma_{xy}^{(0)} + M_{xy} \delta \gamma_{xy}^{(1)} \right. \right. \\ \left. \left. + Q_x \delta \gamma_{xz}^{(0)} + Q_y \delta \gamma_{yz}^{(0)} - q \delta w_0 - I_0 (\dot{u}_0 \delta \dot{u}_0 + \dot{v}_0 \delta \dot{v}_0 + \dot{w}_0 \delta \dot{w}_0) \right. \right. \\ \left. \left. - I_1 (\dot{\phi}_x \delta \dot{u}_0 + \dot{\phi}_y \delta \dot{v}_0 + \delta \dot{\phi}_x \dot{u}_0 + \delta \dot{\phi}_y \dot{v}_0) - I_2 (\dot{\phi}_x \delta \dot{\phi}_x + \dot{\phi}_y \delta \dot{\phi}_y) \right] dx dy \right. \\ \left. - \int_{\Gamma_\sigma} (\hat{N}_{nn} \delta u_n + \hat{N}_{ns} \delta u_s + \hat{M}_{nn} \delta \phi_n + \hat{M}_{ns} \delta \phi_s + \hat{Q}_n \delta w_0) ds \right\} dt \quad (3.4.9)$$

(6.2-8)

## Shear Correction Factor

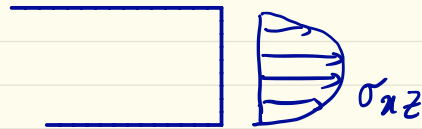
we had 
$$\begin{Bmatrix} Q_x \\ Q_y \end{Bmatrix} = \int_{-h/2}^{h/2} \begin{Bmatrix} \sigma_{xz} \\ \sigma_{yz} \end{Bmatrix} dz \quad (6.2-9)$$

In FSDT we assumed that shear stress is constant through the thickness

$$\sigma_{xz} = \frac{Q_x}{A} \quad (6.2-10)$$

$A \rightarrow$  cross section area

But is well known from elementary theory of homogeneous beam that the transverse shear stress varies parabolically through the beam thickness.



This assumption make a mistake in calculating strain energy. It is often corrected by multiplying with a parameter  $K$ , called shear correction coefficient.

$$\begin{Bmatrix} Q_x \\ Q_y \end{Bmatrix} = K \int_{-h/2}^{h/2} \begin{Bmatrix} \sigma_{xz} \\ \sigma_{yz} \end{Bmatrix} dz \quad (6.2-11)$$

distribution

$$K = \frac{\int U^c}{\int U^F} \rightarrow \begin{matrix} \text{strain energy based on actual shear stress} \\ \text{FSDT} \end{matrix}$$

For example for rectangular cross section the actual shear stress distribution through the thickness is given by

$$\sigma_{xz}^C = \frac{3Q}{2bh} \left[ 1 - \left( \frac{2z}{h} \right)^2 \right] \quad (6.2-13)$$

$$\left\{ \begin{aligned} U^C &= \frac{1}{2G_{13}} \int_A (\sigma_{xz}^C)^2 dA = \frac{3Q^2}{5G_{13}bh} \end{aligned} \right. \quad (6.2-14)$$

$$\left\{ \begin{aligned} U^F &= \frac{1}{2G_{13}} \int_A (\sigma_{xz}^F)^2 dA = \frac{Q^2}{3G_{13}bh} \end{aligned} \right.$$

$$\Rightarrow k = \frac{U^C}{U^F} = \frac{5}{6} \quad (\text{Rectangular})$$