

# Composites

# Lesson 15

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The variational operator ( $\delta$ ) acts on the dependent variables (functionals) in the same way that the differential operator ( $d$ ) acts on the independent variable.

$$\delta(F_1 F_2) = (\delta F_1) F_2 + F_1 (\delta F_2)$$

$$\delta(F_1^2) = 2F_1 (\delta F_1)$$

$$\delta(y^2) = 2y \delta y$$

$$\frac{d}{dt}(\delta y) = \delta\left(\frac{dy}{dt}\right)$$

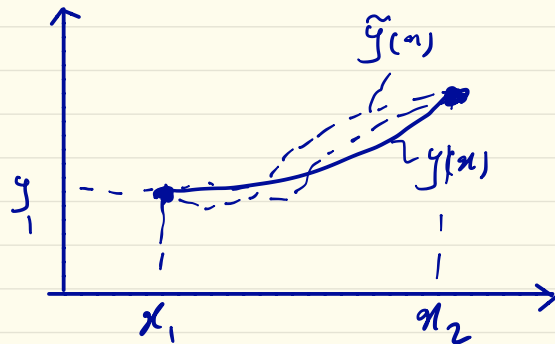
the property of commutativity

Example:

$$y = x^2 + 3 \sin(x)$$

$$\delta(x^2 + 3 \sin(x)) = 0$$

$$\delta(y) \neq 0$$



Example:

$$I(u) = \int_{x_1}^{x_2} \left( \frac{c_0}{2} u^2 + \frac{c_1}{4} \dot{u}^4 + \frac{c_2}{2} \ddot{u}^2 + f(x)u \right) dx$$

$$\begin{cases} y(x_1) = y_1 \\ y(x_2) = y_2 \\ \delta y|_{x=x_i} = 0 \end{cases}$$

$\delta I = ?$

$$\delta I = \int \delta F dx = \int (c_0 u \delta u + c_1 \dot{u}^3 \delta \dot{u} + c_2 \ddot{u} \delta \ddot{u} + \delta f u + f \delta u) dx$$

## 5.2. Boundary Conditions in plates

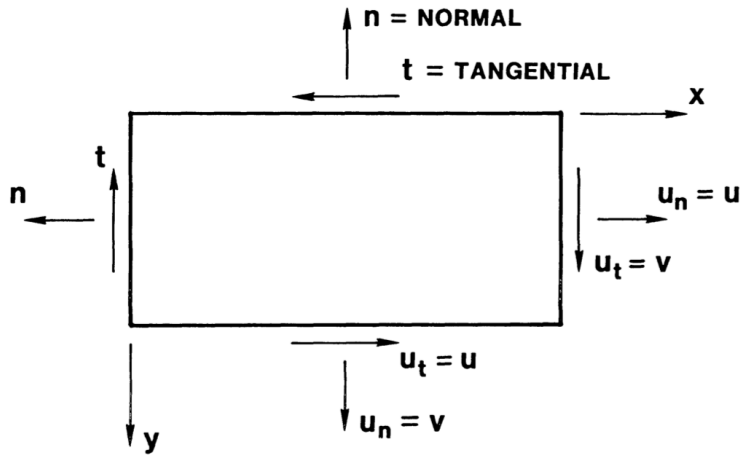
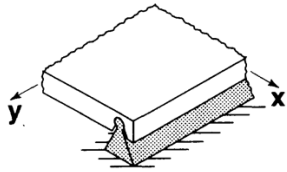


Figure 5-6 Boundary Condition Notation



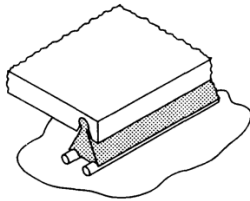
**S1**

$$w = 0$$

$$M_x = 0$$

$$u = 0$$

$$v = 0$$



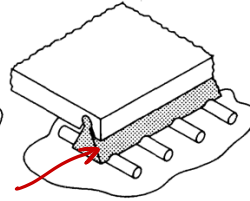
**S2**

$$w = 0$$

$$M_x = 0$$

$$N_x = 0$$

$$v = 0$$



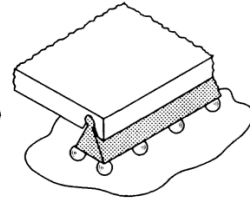
**S3**

$$w = 0$$

$$M_x = 0$$

$$u = 0$$

$$N_{xy} = 0$$



**S4**

$$w = 0$$

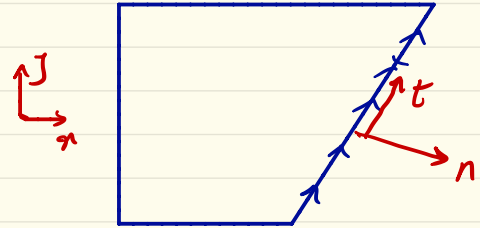
$$M_x = 0$$

$$N_x = 0$$

$$N_{xy} = 0$$

- (1) For the S1 condition, the triangular prism is fastened to a horizontal surface such that the prism cannot translate in either the x- or y-directions. Thus,  $u = 0$  and  $v = 0$ . Hence, forces  $N_x$  and  $N_{xy}$  must exist.
- (2) For the S2 condition, the triangular prism is supported on partially embedded y-direction roller bearings that permit translation in the x-direction, but none in the y-direction. Thus,  $u \neq 0$  and  $v = 0$ . Hence, force  $N_x$  must be zero, but  $N_{xy}$  must exist.
- (3) For the S3 condition, the triangular prism is supported on partially embedded x-direction roller bearings that permit translation in the y-direction, but none in the x-direction. Thus,  $u = 0$  and  $v \neq 0$ . Hence, force  $N_x$  must exist, but  $N_{xy}$  must be zero.
- (4) For the S4 condition, the triangular prism is supported on partially embedded spherical bearings that permit translation in any direction in the x-y plane. Thus,  $u \neq 0$  and  $v \neq 0$ , so both forces  $N_x$  and  $N_{xy}$  must be zero.

S1:	$w = 0$	$M_n = 0$	$u_n = \bar{u}_n$	$u_t = \bar{u}_t$
S2:	$w = 0$	$M_n = 0$	$N_n = \bar{N}_n$	$u_t = \bar{u}_t$
S3:	$w = 0$	$M_n = 0$	$u_n = \bar{u}_n$	$N_{nt} = \bar{N}_{nt}$
S4:	$w = 0$	$M_n = 0$	$N_n = \bar{N}_n$	$N_{nt} = \bar{N}_{nt}$
C1:	$w = 0$	$w_{,n} = 0$	$u_n = \bar{u}_n$	$u_t = \bar{u}_t$
C2:	$w = 0$	$w_{,n} = 0$	$N_n = \bar{N}_n$	$u_t = \bar{u}_t$
C3:	$w = 0$	$w_{,n} = 0$	$u_n = \bar{u}_n$	$N_{nt} = \bar{N}_{nt}$
C4:	$w = 0$	$w_{,n} = 0$	$N_n = \bar{N}_n$	$N_{nt} = \bar{N}_{nt}$



### 5.3. Orthogonal functions in Solving equations

$$\frac{dx}{dx} + 4 \frac{d^2x}{dy^2} + 5x^2 + 6 \sin(x) = 0 \rightsquigarrow L(x) = 0$$

$$L(\dots) = \frac{d}{dy}(\dots) + 4 \frac{d^2}{dy^2}(\dots) + 5(\dots)^2 + 6 \sin(\dots) \quad \text{operator}$$

$$L(x, y) \equiv 5x^2 + 6y^2 + 3xy \quad \text{two-variables operator}$$

u & v are orthogonal based on  $L(x, y)$  if:  $L(u, v) = 0$

dot operation:  $\vec{a} \perp \vec{b}$  if  $\vec{a} \cdot \vec{b} = 0$

$$\frac{1}{\pi} \int_0^{2\pi} \sin(nx) \sin(mx) dx = \begin{cases} 0 & n \neq m \\ 1 & n = m \end{cases}$$

$$\int_0^a \sin\left(\frac{n\pi}{a}x\right) \sin\left(\frac{m\pi}{a}x\right) dx = \begin{cases} 0 & n \neq m \\ \frac{a}{2} & n = m \end{cases}$$

$$P(x) = a_1 \sin\left(\frac{\pi}{a}x\right) + a_2 \sin\left(\frac{2\pi}{a}x\right) + a_3 \sin\left(\frac{3\pi}{a}x\right) + \dots$$

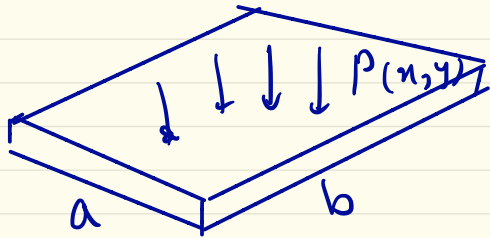
$$\int_0^a P(x) \sin\left(\frac{2\pi}{a}x\right) dx = 0 + a_2 \frac{a}{2} + 0 + \dots$$

$$a_2 = \frac{2}{a} \int_0^a P(x) \sin\left(\frac{2\pi}{a}x\right) dx$$

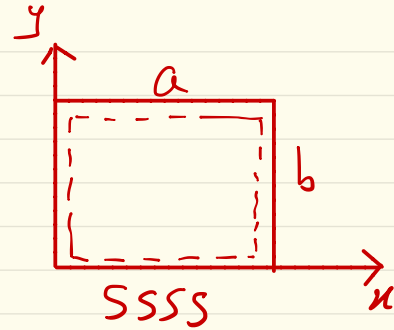
$$a_i = \frac{2}{a} \int_0^a P(x) \sin\left(\frac{i\pi}{a}x\right) dx$$



application example: (Navier approach)



$$\nabla^4 w = \frac{P}{D}$$



$$P = \sum_n \sum_m P_{mn} \sin\left(\frac{m\pi x}{a}\right) \sin\left(\frac{n\pi y}{b}\right)$$

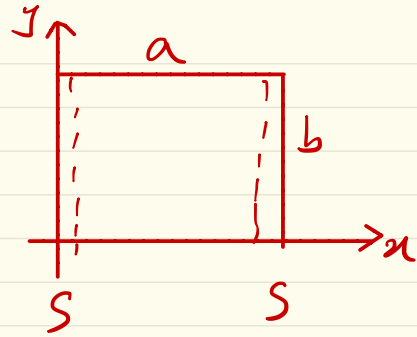
$$w = \sum_n \sum_m w_{mn} \sin\left(\frac{m\pi x}{a}\right) \sin\left(\frac{n\pi y}{b}\right)$$

$$\begin{aligned} w_{mn} \left( \frac{\partial^4}{\partial x^4} + 2 \frac{\partial^4}{\partial x^2 \partial y^2} + \frac{\partial^4}{\partial y^4} \right) \left( \sin\left(\frac{m\pi x}{a}\right) \sin\left(\frac{n\pi y}{b}\right) \right) \\ = \frac{P_{mn}}{D} \sin\left(\frac{m\pi x}{a}\right) \sin\left(\frac{n\pi y}{b}\right) \end{aligned}$$

$$w_{mn} \left( \left(\frac{m}{a}\right)^2 + \left(\frac{n}{b}\right)^2 \right)^2 \cdot \pi^4 = \frac{P_{mn}}{D}$$

application example: (Levy approach)

$$\nabla^4 w = \frac{p}{D}$$



$$\begin{cases} w = \sum_m Y_m(y) \sin \frac{m\pi x}{a} \\ p = \sum_m P_m(y) \sin \frac{m\pi x}{a} \end{cases}$$

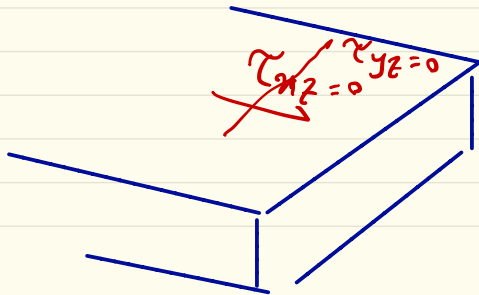
$$\sum_m \left( Y_m^{IV} - 2 \frac{m^2 \pi^2}{a^2} Y_m'' + \frac{m^4 \pi^4}{a^4} Y_m \right) \sin \left( \frac{m\pi x}{a} \right) = \sum_m \frac{P_m(y)}{D} \sin \frac{m\pi x}{a}$$

$$\Rightarrow Y_m^{IV}(y) - 2 \frac{m^2 \pi^2}{a^2} Y_m'' + \frac{m^4 \pi^4}{a^4} Y_m = \frac{P_m(y)}{D}$$

# chapter 6: classical and First-Order Theories of Laminated Composite Plates

## General assumption

- The layers are perfectly bonded together (assumption).
- The material of each layer is linearly elastic and has three planes of material symmetry (i.e., orthotropic) (restriction).
- Each layer is of uniform thickness (restriction).
- The strains and displacements are small (restriction).
- The transverse shear stresses on the top and bottom surfaces of the laminate are zero (restriction).



# 1. Classical laminated plate theory (CLPT)

which is an extension of the Kirchhoff (classical) plate theory

$$u(x, y, z, t) = u_0(x, y, t) - z \frac{\partial w_0}{\partial x}$$

$$v(x, y, z, t) = v_0(x, y, t) - z \frac{\partial w_0}{\partial y}$$

$$w(x, y, z, t) = w_0(x, y, t)$$

where  $(u_0, v_0, w_0)$  are the displacement components along the  $(x, y, z)$  coordinate.

# 2. The First-order Shear deformation Theory (FSDT)

$$u(x, y, z, t) = u_0(x, y, t) + z\phi_x(x, y, t)$$

$$v(x, y, z, t) = v_0(x, y, t) + z\phi_y(x, y, t)$$

$$w(x, y, z, t) = w_0(x, y, t)$$

### 3- Third-order laminated plate Theory of Reddy (TLPT)

$$u(x, y, z, t) = u_0(x, y, t) + z\phi_x(x, y, t) + z^3 \left( -\frac{4}{3h^2} \right) \left( \phi_x + \frac{\partial w_0}{\partial x} \right)$$

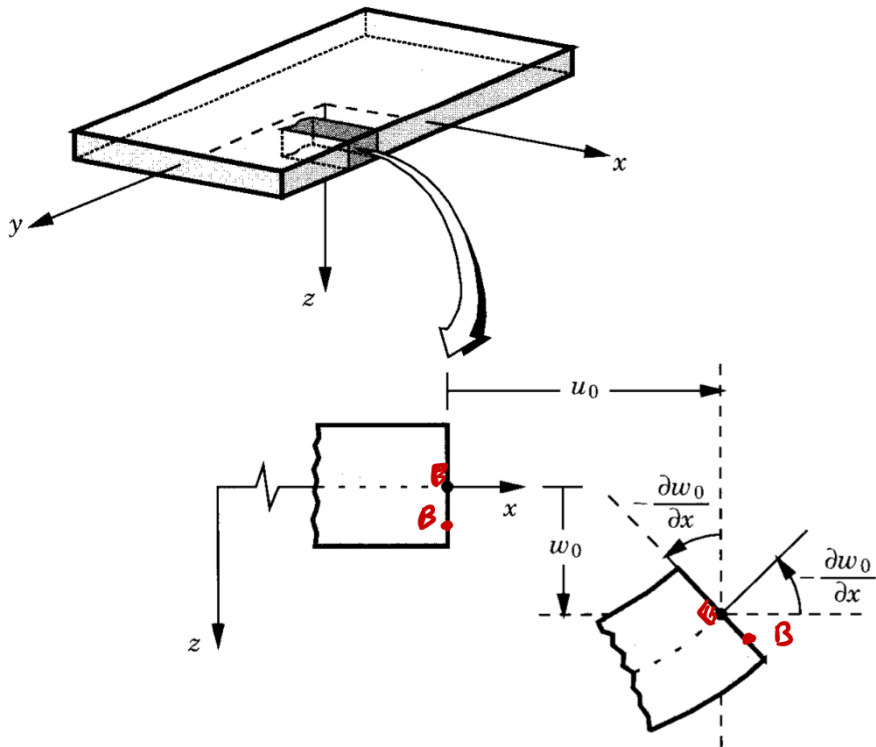
$$v(x, y, z, t) = v_0(x, y, t) + z\phi_y(x, y, t) + z^3 \left( -\frac{4}{3h^2} \right) \left( \phi_y + \frac{\partial w_0}{\partial y} \right)$$

$$w(x, y, z, t) = w_0(x, y, t)$$

## 6.1\_ The classical Laminated Plate Theory (CLPT)

- (1) Straight lines perpendicular to the midsurface (i.e., transverse normals) before deformation remain straight after deformation.
- (2) The transverse normals do not experience elongation (i.e., they are inextensible).
- (3) The transverse normals rotate such that they remain perpendicular to the midsurface after deformation.

$$\epsilon_2 \approx 0$$



## 6.1-1 Displacements and Strains

$$u(x, y, z, t) = u_0(x, y, t) - z \frac{\partial w_0}{\partial x}$$

$$v(x, y, z, t) = v_0(x, y, t) - z \frac{\partial w_0}{\partial y}$$

$$w(x, y, z, t) = w_0(x, y, t)$$