2.5 - Tensile Strength in the Fiber Direction



 Both fibers and matrix defarm elastically
 The fiber continue to defarm elastically but the matrix defarms plastically
 Both the fibers and the matrix defarm plastically
 The fibers fracture followed by fracture of the composite material

More than a certain minimum volume fraction of fibers (24)
ultimate strength of composite is achieved when the fibers
are strained to correspond to their maximmum (ultimate) stress.

$$E_{cmax} = E_{fmax}$$
 (2.5-1)
and we had that
 $\sigma_{cmax} = \sigma_{fmax} V_f + (\sigma_m) \sum_{fmax} (1-V_f)$ (2.5-2)
 F_{max}
 $\sigma_{fmax} = maximum fiber tensik stress
(σ_m) = matrix stress at a matin strain equal
 F_{fmax} to the maximum tensile strain
in the fibers.$



we use fiber reinforcement to lead to a greater Strength than can be obtained wit the matrix alone. $\mathcal{O}_{Cmax} > \mathcal{O}_{mmax}$ (2.5-3)

$$\frac{(2.5.2)f(2.5.3)}{Cr} = \frac{Ommax - (Om)Efmax}{Ofmax} (2.5.4)$$

For very small values of
$$2_{f}$$
, the behaivior of the composite
material might not follow Eq. (2.5-2) because there might not
be enough fibers to control the matrix elonglation. Therefore
all fibers break at the first moment and the
matrix dominates the Composite material
 $\delta_{Cmax} = \delta_{mmax}(1-V_{f})$ (2.5-5)

5 Crax 1 $\overline{\mathcal{O}_{max}} = \overline{\mathcal{O}_{f}} \frac{\mathcal{V}_{f+}(\mathcal{O}_{m})}{\mathcal{E}_{f}} \left(1 - \mathcal{V}_{f}\right)$ Smax $\int_{1}^{1} \int_{2}^{2} \int_{2$ >VF Vmin Vcr $\widetilde{O}_{m}\max(1-\widetilde{V}_{f}) = \widetilde{O}_{f}\max(1-\widetilde{V}_{f}) + (\widetilde{O}_{m})\sum_{f}(1-\widetilde{V}_{f})$ Ommax - (Om)Efmax $v_{min} =$ ofmax + 6 mmax (6 m) Efmax

$$V_{f} \leqslant v_{min} \Rightarrow$$
 The Composite Strength is controlled by the matrix
 $V_{f} \leqslant v_{cr} \Rightarrow$ The Composite Strength is controlled by the fiber deform
but the purpose of producing composites has not
satisfied
 $V \leqslant v_{f} = 0$ only for this case the Composite material gain
strength from howing fiber reinforcement.



b: Symmetric. Horizontal lines of original length 2C extend or contract. Thus, the matrix extends or contracts in y-direction transverse to the X-direction.

C: Antisymmetric. The matrix is subjected to shearing deformation Horizontal lines. do not change orientation or length.

TransversOr Extensionmode $\sigma_{f_{cr}} = 2\sqrt{\frac{V_f E_m E_f}{3(1-V_f)}}$ $\sigma_{c_{max}} = 2\left[V_f + (1-V_f)\frac{E_m}{E_f}\right]\sqrt{\frac{V_f E_m E_f}{3(1-V_f)}}$ (2.6-1)Shearmode $\sigma_{c_{max}} = \frac{G_m}{1-V_f}$ (2.6-1) $\sigma_{f_{cr}} = \frac{G_m}{V_f(1-V_f)}$ $\sigma_{c_{max}} = \frac{G_m}{1-V_f}$ (2.6-2)



chapter 3_ Macromecanical Behaivior of a Lamina

 $\mathcal{O} = \mathcal{O}_{ij} = \begin{bmatrix} \mathcal{O}_{11} & \mathcal{O}_{12} & \mathcal{O}_{13} \\ \mathcal{O}_{21} & \mathcal{O}_{22} & \mathcal{O}_{23} \\ \mathcal{O}_{31} & \mathcal{O}_{32} & \mathcal{O}_{33} \\ \mathcal{O}_{31} & \mathcal{O}_{32} & \mathcal{O}_{33} \\ \mathcal{O}_{23} \end{bmatrix} \xrightarrow{\mathcal{O}_{12}}_{\mathcal{O}_{23}}$ > 0 (2) Quadratic tensor

Cauchy's stress thearem

