

chapter VII Third-order Theory of Laminated Composite Plates

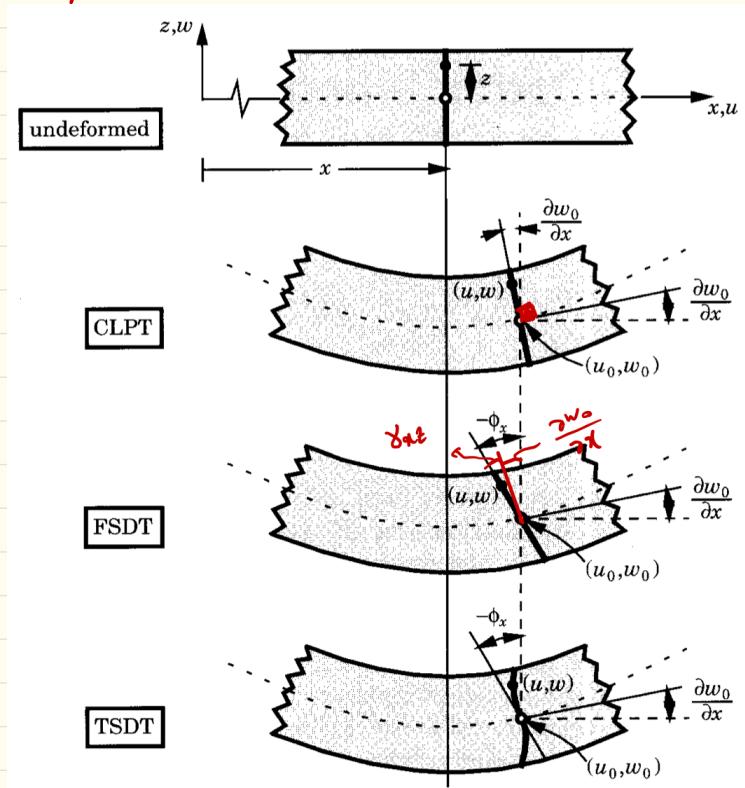
7.1 - Introduction

تئوری هار درجه بالا سینماتیک مانند رایتھر مدل می کند راحتیا جم به ضریب اصلاح برجی (K) نظریه همینیستی بین لایه ای رایتھر باید می کند ایا هزینه محاسبات بیشتری دارند لذا باید رمث تئوری وقتی نیاز است از این تئوری حماستن آن شود.

علیکه توزیع جایگائی در جهت مقاومت درجه ۳ در تئوری کرفته شده است این است که بتوان به توزیع هست برجی درجه ۲ در مقاومت رسید که بر واقعیت نزدیک است. لذا در تئوری کرمی در جای بالاتر بر این توزیع جایگائی در مقاومت بسته مورد نیاز نیست.

7.2 A Third-order Plate Theory

7.2-1 Displacement Field



$$\begin{aligned} u &= u_0 + z\phi_x + z^2\theta_x + z^3\lambda_x \\ v &= v_0 + z\phi_x + z^2\theta_y + z^3\lambda_y \\ w &= w_0 \end{aligned}$$

(7.2-1)

$$\begin{aligned} u_0 &= u(x, y, 0, t), & v_0 &= v(x, y, 0, t), & w_0 &= w(x, y, 0, t) \\ \phi_x &= \left(\frac{\partial u}{\partial z} \right)_{z=0}, & \phi_y &= \left(\frac{\partial v}{\partial z} \right)_{z=0} & 2\theta_x &= \left(\frac{\partial^2 u}{\partial z^2} \right)_{z=0}, \\ 2\theta_y &= \left(\frac{\partial^2 v}{\partial z^2} \right)_{z=0} & 6\lambda_x &= \left(\frac{\partial^3 u}{\partial z^3} \right)_{z=0}, & 6\lambda_y &= \left(\frac{\partial^3 v}{\partial z^3} \right)_{z=0} \end{aligned}$$

(7.2-2)

آگر دنفر تابه کو در صورت بالا دایسیں تئی بری خارجی نداشتے بازم، می توانیں لفٹے:

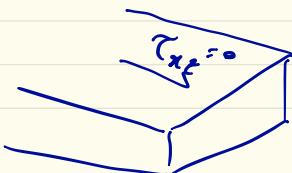
$$\sigma_{xz}(x, y, \pm h/2, t) = 0, \quad \sigma_{yz}(x, y, \pm h/2, t) = 0$$

(7.2-3)

\Rightarrow

$$\begin{aligned} 0 &= \sigma_{xz}(x, y, \pm h/2, t) = Q_{55}\gamma_{xz}(x, y, \pm h/2, t) + Q_{45}\gamma_{yz}(x, y, \pm h/2, t), \\ 0 &= \sigma_{yz}(x, y, \pm h/2, t) = Q_{45}\gamma_{xz}(x, y, \pm h/2, t) + Q_{44}\gamma_{yz}(x, y, \pm h/2, t) \end{aligned}$$

(7.2-4)



$$\begin{aligned}\phi_x + \frac{\partial w_0}{\partial x} + \left(-h\theta_x + \frac{3h^2}{4}\lambda_x \right) &= 0, & \phi_x + \frac{\partial w_0}{\partial x} + \left(h\theta_x + \frac{3h^2}{4}\lambda_x \right) &= 0 \\ \phi_y + \frac{\partial w_0}{\partial y} + \left(-h\theta_y + \frac{3h^2}{4}\lambda_y \right) &= 0, & \phi_y + \frac{\partial w_0}{\partial y} + \left(h\theta_y + \frac{3h^2}{4}\lambda_y \right) &= 0\end{aligned}$$

or

$$\lambda_x = -\frac{4}{3h^2} \left(\phi_x + \frac{\partial w_0}{\partial x} \right), \quad \theta_x = 0; \quad \lambda_y = -\frac{4}{3h^2} \left(\phi_y + \frac{\partial w_0}{\partial y} \right), \quad \theta_y = 0$$

(٧.٢-٥)

=>

$$\begin{aligned}u(x, y, z, t) &= u_0(x, y, t) + z\phi_x(x, y, t) - \frac{4}{3h^2}z^3 \left(\phi_x + \frac{\partial w_0}{\partial x} \right) \\ v(x, y, z, t) &= v_0(x, y, t) + z\phi_y(x, y, t) - \frac{4}{3h^2}z^3 \left(\phi_y + \frac{\partial w_0}{\partial y} \right) \\ w(x, y, z, t) &= w_0(x, y, t)\end{aligned}$$

c_1

TSDT
ردیغ بدن سی بری

(٧.٢-٦)

7.2-2 Strains

$$\begin{Bmatrix} \varepsilon_{xx} \\ \varepsilon_{yy} \\ \gamma_{xy} \end{Bmatrix} = \begin{Bmatrix} \varepsilon_{xx}^{(0)} \\ \varepsilon_{yy}^{(0)} \\ \gamma_{xy}^{(0)} \end{Bmatrix} + z \begin{Bmatrix} \varepsilon_{xx}^{(1)} \\ \varepsilon_{yy}^{(1)} \\ \gamma_{xy}^{(1)} \end{Bmatrix} + z^3 \begin{Bmatrix} \varepsilon_{xx}^{(3)} \\ \varepsilon_{yy}^{(3)} \\ \gamma_{xy}^{(3)} \end{Bmatrix}$$

$$\begin{Bmatrix} \gamma_{yz} \\ \gamma_{xz} \end{Bmatrix} = \begin{Bmatrix} \gamma_{yz}^{(0)} \\ \gamma_{xz}^{(0)} \end{Bmatrix} + z^2 \begin{Bmatrix} \gamma_{yz}^{(2)} \\ \gamma_{xz}^{(2)} \end{Bmatrix}$$

(7.2-7)

$$C_1 \rightarrow C_2 = 3C_1 \rightarrow C_1 = \frac{4}{3h^2} \text{ با خزن}$$

$$\begin{Bmatrix} \varepsilon_{xx}^{(0)} \\ \varepsilon_{yy}^{(0)} \\ \gamma_{xy}^{(0)} \end{Bmatrix} = \begin{Bmatrix} \frac{\partial u_0}{\partial x} + \frac{1}{2} \left(\frac{\partial w_0}{\partial x} \right)^2 \\ \frac{\partial v_0}{\partial y} + \frac{1}{2} \left(\frac{\partial w_0}{\partial y} \right)^2 \\ \frac{\partial u_0}{\partial y} + \frac{\partial v_0}{\partial x} + \frac{\partial w_0}{\partial x} \frac{\partial w_0}{\partial y} \end{Bmatrix}$$

$$\begin{Bmatrix} \varepsilon_{xx}^{(1)} \\ \varepsilon_{yy}^{(1)} \\ \gamma_{xy}^{(1)} \end{Bmatrix} = \begin{Bmatrix} \frac{\partial \phi_x}{\partial x} \\ \frac{\partial \phi_y}{\partial y} \\ \frac{\partial \phi_x}{\partial y} + \frac{\partial \phi_y}{\partial x} \end{Bmatrix}, \quad \begin{Bmatrix} \gamma_{yz}^{(0)} \\ \gamma_{xz}^{(0)} \end{Bmatrix} = \begin{Bmatrix} \phi_y + \frac{\partial w_0}{\partial y} \\ \phi_x + \frac{\partial w_0}{\partial x} \end{Bmatrix}$$

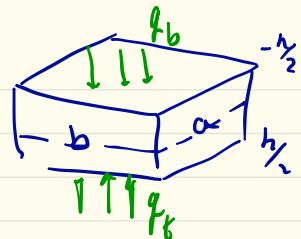
$$\begin{Bmatrix} \varepsilon_{xx}^{(3)} \\ \varepsilon_{yy}^{(3)} \\ \gamma_{xy}^{(3)} \end{Bmatrix} = -c_1 \begin{Bmatrix} \frac{\partial \phi_x}{\partial x} + \frac{\partial^2 w_0}{\partial x^2} \\ \frac{\partial \phi_y}{\partial y} + \frac{\partial^2 w_0}{\partial y^2} \\ \frac{\partial \phi_x}{\partial y} + \frac{\partial \phi_y}{\partial x} + 2 \frac{\partial^2 w_0}{\partial x \partial y} \end{Bmatrix}, \quad \begin{Bmatrix} \gamma_{yz}^{(2)} \\ \gamma_{xz}^{(2)} \end{Bmatrix} = -c_2 \begin{Bmatrix} \phi_y + \frac{\partial w_0}{\partial y} \\ \phi_x + \frac{\partial w_0}{\partial x} \end{Bmatrix}$$

(7.2-8)

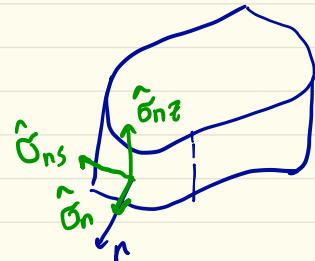
7.2 - 3 Equation of Motion

$$\begin{aligned}\delta U &= \int_{\Omega_0} \left\{ \int_{-\frac{h}{2}}^{\frac{h}{2}} \left[\sigma_{xx} (\delta \varepsilon_{xx}^{(0)} + z \delta \varepsilon_{xx}^{(1)} - c_1 z^3 \delta \varepsilon_{xx}^{(3)}) \right. \right. \\ &\quad + \sigma_{yy} (\delta \varepsilon_{yy}^{(0)} + z \delta \varepsilon_{yy}^{(1)} - c_1 z^3 \delta \varepsilon_{yy}^{(3)}) + \sigma_{xy} (\delta \gamma_{xy}^{(0)} + z \delta \gamma_{xy}^{(1)} - c_1 z^3 \delta \gamma_{xy}^{(3)}) \\ &\quad \left. \left. + \sigma_{xz} (\delta \gamma_{xz}^{(0)} + z^2 \delta \gamma_{xz}^{(2)}) + \sigma_{yz} (\delta \gamma_{yz}^{(0)} + z^2 \delta \gamma_{yz}^{(2)}) \right] dz \right\} dx dy \\ &= \int_{\Omega_0} \left(N_{xx} \delta \varepsilon_{xx}^{(0)} + M_{xx} \delta \varepsilon_{xx}^{(1)} - c_1 P_{xx} \delta \varepsilon_{xx}^{(3)} + N_{yy} \delta \varepsilon_{yy}^{(0)} + M_{yy} \delta \varepsilon_{yy}^{(1)} - c_1 P_{yy} \delta \varepsilon_{yy}^{(3)} \right. \\ &\quad \left. + N_{xy} \delta \gamma_{xy}^{(0)} + M_{xy} \delta \gamma_{xy}^{(1)} - c_1 P_{xy} \delta \gamma_{xy}^{(3)} \right. \\ &\quad \left. + Q_x \delta \gamma_{xz}^{(0)} - c_2 R_x \delta \gamma_{xz}^{(2)} + Q_y \delta \gamma_{yz}^{(0)} - c_2 R_y \delta \gamma_{yz}^{(2)} \right) dx dy\end{aligned}$$

$$\begin{aligned}\delta V &= - \int_{\Omega_0} \left[q_b(x, y) \delta w(x, y, -\frac{h}{2}) + q_t(x, y) \delta w(x, y, \frac{h}{2}) \right] dx dy \\ &\quad - \int_{\Gamma} \int_{-\frac{h}{2}}^{\frac{h}{2}} \left[\hat{\sigma}_{nn} (\delta u_n + z \delta \phi_n - c_1 z^3 \delta \varphi_n) \right. \\ &\quad \left. + \hat{\sigma}_{ns} (\delta u_s + z \delta \phi_s - c_1 z^3 \delta \varphi_{ns}) + \hat{\sigma}_{nz} \delta w_0 \right] dz d\Gamma \\ &= - \int_{\Omega_0} q \delta w_0 dx dy - \int_{\Gamma} (\hat{N}_{nn} \delta u_n + \hat{M}_{nn} \delta \phi_n - c_1 \hat{P}_{nn} \delta \varphi_n \\ &\quad + \hat{N}_{ns} \delta u_s + \hat{M}_{ns} \delta \phi_s - c_1 \hat{P}_{ns} \delta \varphi_{ns} + \hat{Q}_n \delta w_0) d\Gamma\end{aligned}$$



(7.2-9)



(7.2-10)

$$\begin{aligned}
\delta K &= \int_{\Omega_0} \int_{-\frac{h}{2}}^{\frac{h}{2}} \rho_0 \left[\left(\dot{u}_0 + z\dot{\phi}_x - c_1 z^3 \dot{\varphi}_x \right) \left(\delta \dot{u}_0 + z\delta\dot{\phi}_x - c_1 z^3 \delta\dot{\varphi}_x \right) \right. \\
&\quad \left. + \left(\dot{v}_0 + z\dot{\phi}_y - c_1 z^3 \dot{\varphi}_y \right) \left(\delta \dot{v}_0 + z\delta\dot{\phi}_y - c_1 z^3 \delta\dot{\varphi}_y \right) + \dot{w}_0 \delta \dot{w}_0 \right] dv \\
&= \int_{\Omega_0} \left[\left(I_0 \dot{u}_0 + I_1 \dot{\phi}_x - c_1 I_3 \dot{\varphi}_x \right) \delta \dot{u}_0 + \left(I_1 \dot{u}_0 + I_2 \dot{\phi}_x - c_1 I_4 \dot{\varphi}_x \right) \delta \dot{\phi}_x \right. \\
&\quad - c_1 \left(I_3 \dot{u}_0 + I_4 \dot{\phi}_x - c_1 I_6 \dot{\varphi}_x \right) \delta \dot{\varphi}_x + \left(I_0 \dot{v}_0 + I_1 \dot{\phi}_y - c_1 I_3 \dot{\varphi}_y \right) \delta \dot{v}_0 \\
&\quad \left. + \left(I_1 \dot{v}_0 + I_2 \dot{\phi}_y - c_1 I_4 \dot{\varphi}_y \right) \delta \dot{\phi}_y - c_1 \left(I_3 \dot{u}_0 + I_4 \dot{\phi}_y - c_1 I_6 \dot{\varphi}_y \right) \delta \dot{\varphi}_y \right] dx dy
\end{aligned}$$

لایه مانع اس . سمبیتی

$$\begin{Bmatrix} N_{\alpha\beta} \\ M_{\alpha\beta} \\ P_{\alpha\beta} \end{Bmatrix} = \int_{-\frac{h}{2}}^{\frac{h}{2}} \sigma_{\alpha\beta} \begin{Bmatrix} 1 \\ z \\ z^3 \end{Bmatrix} dz, \quad \begin{Bmatrix} Q_\alpha \\ R_\alpha \end{Bmatrix} = \int_{-\frac{h}{2}}^{\frac{h}{2}} \sigma_{\alpha z} \begin{Bmatrix} 1 \\ z^2 \end{Bmatrix} dz$$

(Z.2-12)

$$I_i = \int_{-\frac{h}{2}}^{\frac{h}{2}} \rho_0(z)^i dz \quad (i = 0, 1, 2, \dots, 6)$$

۱۶ استاد از روابط فوق معارض حکم دینی می‌گوند.

$$\frac{\partial N_{xx}}{\partial x} + \frac{\partial N_{xy}}{\partial y} = I_0 \ddot{u}_0 + J_1 \ddot{\phi}_x - c_1 I_3 \frac{\partial \ddot{w}_0}{\partial x}$$

a

$$\frac{\partial N_{xy}}{\partial x} + \frac{\partial N_{yy}}{\partial y} = I_0 \ddot{v}_0 + J_1 \ddot{\phi}_y - c_1 I_3 \frac{\partial \ddot{w}_0}{\partial y}$$

b

$$\frac{\partial \bar{Q}_x}{\partial x} + \frac{\partial \bar{Q}_y}{\partial y} + \frac{\partial}{\partial x} (N_{xx} \frac{\partial w_0}{\partial x} + N_{xy} \frac{\partial w_0}{\partial y}) + \frac{\partial}{\partial y} (N_{xy} \frac{\partial w_0}{\partial x} + N_{yy} \frac{\partial w_0}{\partial y})$$

(۷.۲-۱۳)

$$+ c_1 \left(\frac{\partial^2 P_{xx}}{\partial x^2} + 2 \frac{\partial^2 P_{xy}}{\partial x \partial y} + \frac{\partial^2 P_{yy}}{\partial y^2} \right) + q = I_0 \ddot{w}_0 - c_1^2 I_6 \left(\frac{\partial^2 \ddot{w}_0}{\partial x^2} + \frac{\partial^2 \ddot{w}_0}{\partial y^2} \right)$$

c

$$+ c_1 \left[I_3 \left(\frac{\partial \ddot{u}_0}{\partial x} + \frac{\partial \ddot{v}_0}{\partial y} \right) + J_4 \left(\frac{\partial \ddot{\phi}_x}{\partial x} + \frac{\partial \ddot{\phi}_y}{\partial y} \right) \right]$$

$$\frac{\partial \bar{M}_{xx}}{\partial x} + \frac{\partial \bar{M}_{xy}}{\partial y} - \bar{Q}_x = J_1 \ddot{u}_0 + K_2 \ddot{\phi}_x - c_1 J_4 \frac{\partial \ddot{w}_0}{\partial x}$$

d

$$\frac{\partial \bar{M}_{xy}}{\partial x} + \frac{\partial \bar{M}_{yy}}{\partial y} - \bar{Q}_y = J_1 \ddot{v}_0 + K_2 \ddot{\phi}_y - c_1 J_4 \frac{\partial \ddot{w}_0}{\partial y}$$

e

$$\bar{M}_{\alpha\beta} = M_{\alpha\beta} - c_1 P_{\alpha\beta} \quad (\alpha, \beta = 1, 2, 6); \quad \bar{Q}_\alpha = Q_\alpha - c_2 R_\alpha \quad (\alpha = 4, 5)$$

ج، ج، ج

$$I_i = \sum_{k=1}^N \int_{z_k}^{z_{k+1}} \rho^{(k)}(z)^i dz \quad (i = 0, 1, 2, \dots, 6)$$

(۷.۲-۱۴)

$$J_i = I_i - c_1 I_{i+2}, \quad K_2 = I_2 - 2c_1 I_4 + c_1^2 I_6, \quad c_1 = \frac{4}{3h^2}, \quad c_2 = 3c_1$$

Primary Variables : $u_n, u_s, w_0, \frac{\partial w_0}{\partial n}, \phi_n, \phi_s$
 Secondary Variables : $N_{nn}, N_{ns}, \bar{V}_n, P_{nn}, \bar{M}_{nn}, \bar{M}_{ns}$

(7.2 - 15)

جواب

$$\begin{aligned} \bar{V}_n &\equiv c_1 \left[\left(\frac{\partial P_{xx}}{\partial x} + \frac{\partial P_{xy}}{\partial y} \right) n_x + \left(\frac{\partial P_{xy}}{\partial x} + \frac{\partial P_{yy}}{\partial y} \right) n_y \right] \\ &\quad - c_1 \left[\left(I_3 \ddot{u}_0 + J_4 \ddot{\phi}_x - c_1 I_6 \frac{\partial \ddot{w}_0}{\partial x} \right) n_x + \left(I_3 \ddot{v}_0 + J_4 \ddot{\phi}_y - c_1 I_6 \frac{\partial \ddot{w}_0}{\partial y} \right) n_y \right] \\ &\quad + (\bar{Q}_x n_x + \bar{Q}_y n_y) + \mathcal{P}(w_0) + c_1 \frac{\partial P_{ns}}{\partial s} \\ \mathcal{P}(w_0) &= \left(N_{xx} \frac{\partial w_0}{\partial x} + N_{xy} \frac{\partial w_0}{\partial y} \right) n_x + \left(N_{xy} \frac{\partial w_0}{\partial x} + N_{yy} \frac{\partial w_0}{\partial y} \right) n_y \end{aligned}$$

(7.2 - 16)

=>

$$\begin{cases} \{N\} \\ \{M\} \\ \{P\} \end{cases} = \begin{bmatrix} [A] & [B] & [E] \\ [B] & [D] & [F] \\ [E] & [F] & [H] \end{bmatrix} \begin{cases} \{\varepsilon^{(0)}\} \\ \{\varepsilon^{(1)}\} \\ \{\varepsilon^{(3)}\} \end{cases}$$

$$\begin{cases} \{Q\} \\ \{R\} \end{cases} = \begin{bmatrix} [A] & [D] \\ [D] & [F] \end{bmatrix} \begin{cases} \{\gamma^{(0)}\} \\ \{\gamma^{(2)}\} \end{cases}$$

معارلہ فنا ی حالت
 TSDT (7.2 - 17)

$$(A_{ij}, B_{ij}, D_{ij}, E_{ij}, F_{ij}, H_{ij}) = \sum_{k=1}^N \int_{z_k}^{z_{k+1}} \bar{Q}_{ij}^{(k)} (1, z, z^2, z^3, z^4, z^6) dz$$

$i, j = 1, 2, 6$

ماتریس حایی

| 7.2-18 |

$$(A_{ij}, D_{ij}, F_{ij}) = \sum_{k=1}^N \int_{z_k}^{z_{k+1}} \bar{Q}_{ij}^{(k)} (1, z^2, z^4) dz$$

$i, j = 1, 2, 5$

ماتریس مارک

| 7.2-19 |

ماتریس حایی A, B, D, F را که قبلاً می‌ستانم. ای ای ماتریس‌های H در فرآیند حین خواهد بود.

$$E_{ij} = \frac{1}{4} \sum_{k=1}^N \bar{Q}_{ij}^{(k)} \left[(z_{k+1})^4 - (z_k)^4 \right]$$

(7.2-20)

$$F_{ij} = \frac{1}{5} \sum_{k=1}^N \bar{Q}_{ij}^{(k)} \left[(z_{k+1})^5 - (z_k)^5 \right]$$

$$H_{ij} = \frac{1}{7} \sum_{k=1}^N \bar{Q}_{ij}^{(k)} \left[(z_{k+1})^7 - (z_k)^7 \right]$$

در معادلات حاکم فوق (7.2-13) با مردادن $C=0$ به معادلات FS-DT

خواهیم رسید.

7.3 Higher-order Laminate Stiffness characteristics

کاھ، تعدد درج سگراھنی ساده سازی می کنند که از P_{xx}, P_{yy}, P_{xy} مرتفع می کنند
لای را رترنفری کرند.

مازی ستی حرالا ی مانند قبل خواهد بود.

$$Q_{11} = \frac{E_1}{1 - \nu_{12}\nu_{21}}, \quad Q_{12} = \frac{\nu_{12}E_2}{1 - \nu_{12}\nu_{21}}, \quad Q_{22} = \frac{E_2}{1 - \nu_{12}\nu_{21}} \quad (7.2-21)$$
$$Q_{66} = G_{12}, \quad Q_{44} = G_{23}, \quad Q_{55} = G_{13}$$

I. Single-Layer Plates

Single Isotropic Layer

$$\begin{cases} N_{xx} \\ N_{yy} \\ N_{xy} \end{cases} = \begin{bmatrix} A_{11} & \nu A_{11} & 0 \\ \nu A_{11} & A_{11} & 0 \\ 0 & 0 & \frac{1-\nu}{2} A_{11} \end{bmatrix} \begin{cases} \varepsilon_{xx}^{(0)} \\ \varepsilon_{yy}^{(0)} \\ \gamma_{xy}^{(0)} \end{cases}$$

$$\begin{cases} M_{xx} \\ M_{yy} \\ M_{xy} \end{cases} = \begin{bmatrix} D_{11} & \nu D_{11} & 0 \\ \nu D_{11} & D_{11} & 0 \\ 0 & 0 & \frac{1-\nu}{2} D_{11} \end{bmatrix} \begin{cases} \varepsilon_{xx}^{(1)} \\ \varepsilon_{yy}^{(1)} \\ \gamma_{xy}^{(1)} \end{cases} + \begin{bmatrix} F_{11} & \nu F_{11} & 0 \\ \nu F_{11} & F_{11} & 0 \\ 0 & 0 & \frac{1-\nu}{2} F_{11} \end{bmatrix} \begin{cases} \varepsilon_{xx}^{(3)} \\ \varepsilon_{yy}^{(3)} \\ \gamma_{xy}^{(3)} \end{cases} \quad (11.3.4b)$$

$$\begin{cases} P_{xx} \\ P_{yy} \\ P_{xy} \end{cases} = \begin{bmatrix} F_{11} & \nu F_{11} & 0 \\ \nu F_{11} & F_{11} & 0 \\ 0 & 0 & \frac{1-\nu}{2} F_{11} \end{bmatrix} \begin{cases} \varepsilon_{xx}^{(1)} \\ \varepsilon_{yy}^{(1)} \\ \gamma_{xy}^{(1)} \end{cases} + \begin{bmatrix} H_{11} & \nu H_{11} & 0 \\ \nu H_{11} & H_{11} & 0 \\ 0 & 0 & \frac{1-\nu}{2} H_{11} \end{bmatrix} \begin{cases} \varepsilon_{xx}^{(3)} \\ \varepsilon_{yy}^{(3)} \\ \gamma_{xy}^{(3)} \end{cases}$$

$$\begin{cases} Q_y \\ Q_x \end{cases} = \frac{1-\nu}{2} \begin{bmatrix} A_{11} & 0 \\ 0 & A_{11} \end{bmatrix} \begin{cases} \gamma_{yz}^{(0)} \\ \gamma_{xz}^{(0)} \end{cases} + \frac{1-\nu}{2} \begin{bmatrix} D_{11} & 0 \\ 0 & D_{11} \end{bmatrix} \begin{cases} \gamma_{yz}^{(2)} \\ \gamma_{xz}^{(2)} \end{cases}$$

$$\begin{cases} R_y \\ R_x \end{cases} = \frac{1-\nu}{2} \begin{bmatrix} D_{11} & 0 \\ 0 & D_{11} \end{bmatrix} \begin{cases} \gamma_{yz}^{(0)} \\ \gamma_{xz}^{(0)} \end{cases} + \frac{1-\nu}{2} \begin{bmatrix} F_{11} & 0 \\ 0 & F_{11} \end{bmatrix} \begin{cases} \gamma_{yz}^{(2)} \\ \gamma_{xz}^{(2)} \end{cases}$$

(7.2-22)

Single Specially Orthotropic Layer

رایمہ N, M, نے مل جو اہدیوں.

$$\begin{Bmatrix} P_{xx} \\ P_{yy} \\ P_{xy} \end{Bmatrix} = \frac{h^5}{80} \begin{bmatrix} Q_{11} & Q_{12} & 0 \\ Q_{12} & Q_{22} & 0 \\ 0 & 0 & Q_{66} \end{bmatrix} \begin{Bmatrix} \varepsilon_{xx}^{(1)} \\ \varepsilon_{yy}^{(1)} \\ \gamma_{xy}^{(1)} \end{Bmatrix} + \frac{h^7}{448} \begin{bmatrix} Q_{11} & Q_{12} & 0 \\ Q_{12} & Q_{22} & 0 \\ 0 & 0 & Q_{66} \end{bmatrix} \begin{Bmatrix} \varepsilon_{xx}^{(3)} \\ \varepsilon_{yy}^{(3)} \\ \gamma_{xy}^{(3)} \end{Bmatrix} \quad (7.2-23)$$

$$\begin{Bmatrix} R_y \\ R_x \end{Bmatrix} = \frac{h^3}{12} \begin{bmatrix} Q_{44} & 0 \\ 0 & Q_{55} \end{bmatrix} \begin{Bmatrix} \gamma_{yz}^{(0)} \\ \gamma_{xz}^{(0)} \end{Bmatrix} + \frac{h^5}{80} \begin{bmatrix} Q_{44} & 0 \\ 0 & Q_{55} \end{bmatrix} \begin{Bmatrix} \gamma_{yz}^{(2)} \\ \gamma_{xz}^{(2)} \end{Bmatrix} \quad (11.3)$$

Single Generally Orthotropic Layer

$$\begin{Bmatrix} P_{xx} \\ P_{yy} \\ P_{xy} \end{Bmatrix} = \begin{bmatrix} F_{11} & F_{12} & F_{16} \\ F_{12} & F_{22} & F_{26} \\ F_{16} & F_{26} & F_{66} \end{bmatrix} \begin{Bmatrix} \varepsilon_{xx}^{(1)} \\ \varepsilon_{yy}^{(1)} \\ \gamma_{xy}^{(1)} \end{Bmatrix} + \begin{bmatrix} H_{11} & H_{12} & H_{16} \\ H_{12} & H_{22} & H_{26} \\ H_{16} & H_{26} & H_{66} \end{bmatrix} \begin{Bmatrix} \varepsilon_{xx}^{(3)} \\ \varepsilon_{yy}^{(3)} \\ \gamma_{xy}^{(3)} \end{Bmatrix}$$

(7.2-24)

$$\begin{Bmatrix} R_y \\ R_x \end{Bmatrix} = \begin{bmatrix} D_{44} & D_{45} \\ D_{45} & D_{55} \end{bmatrix} \begin{Bmatrix} \gamma_{yz}^{(0)} \\ \gamma_{xz}^{(0)} \end{Bmatrix} + \begin{bmatrix} F_{44} & F_{45} \\ F_{45} & F_{55} \end{bmatrix} \begin{Bmatrix} \gamma_{yz}^{(2)} \\ \gamma_{xz}^{(2)} \end{Bmatrix}$$

higher-order thermal stress resultants for this case are given by

$$\begin{Bmatrix} P_{xx}^T \\ P_{yy}^T \\ P_{xy}^T \end{Bmatrix} = \sum_{k=1}^L \int_{z_k}^{z_{k+1}} \begin{bmatrix} \bar{Q}_{11}^k & \bar{Q}_{12}^k & \bar{Q}_{16}^k \\ \bar{Q}_{12}^k & \bar{Q}_{22}^k & \bar{Q}_{26}^k \\ \bar{Q}_{16}^k & \bar{Q}_{26}^k & \bar{Q}_{66}^k \end{bmatrix}^{(k)} \begin{Bmatrix} \alpha_{xx} \\ \alpha_{yy} \\ 2\alpha_{xy} \end{Bmatrix}^{(k)} \Delta T z^3 dz$$

(7.2-25)

II Symmetric Laminates

ماتریس م تابعی (7.2-22) حواهید.

Symmetric Laminates with Multiple Isotropic Layers

$$\begin{Bmatrix} P_{xx} \\ P_{yy} \\ P_{xy} \end{Bmatrix} = \begin{bmatrix} F_{11} & F_{12} & 0 \\ F_{12} & F_{11} & 0 \\ 0 & 0 & F_{66} \end{bmatrix} \begin{Bmatrix} \varepsilon_{xx}^{(1)} \\ \varepsilon_{yy}^{(1)} \\ \gamma_{xy}^{(1)} \end{Bmatrix} + \begin{bmatrix} H_{11} & H_{12} & 0 \\ H_{12} & H_{11} & 0 \\ 0 & 0 & H_{66} \end{bmatrix} \begin{Bmatrix} \varepsilon_{xx}^{(3)} \\ \varepsilon_{yy}^{(3)} \\ \gamma_{xy}^{(3)} \end{Bmatrix} \quad (7.2-26)$$

$$\begin{Bmatrix} R_y \\ R_x \end{Bmatrix} = \begin{bmatrix} D_{44} & 0 \\ 0 & D_{55} \end{bmatrix} \begin{Bmatrix} \gamma_{yz}^{(0)} \\ \gamma_{xz}^{(0)} \end{Bmatrix} + \begin{bmatrix} F_{44} & 0 \\ 0 & F_{55} \end{bmatrix} \begin{Bmatrix} \gamma_{yz}^{(2)} \\ \gamma_{xz}^{(2)} \end{Bmatrix}$$

III Antisymmetric Laminates

$$F_{16} = F_{26} = H_{16} = H_{26} = 0 \quad \& \quad B_{ij}, F_{ij} \neq 0$$

$$\begin{Bmatrix} P_{xx} \\ P_{yy} \\ P_{xy} \end{Bmatrix} = \begin{bmatrix} E_{11} & E_{12} & E_{16} \\ E_{12} & E_{22} & E_{26} \\ E_{16} & E_{26} & E_{66} \end{bmatrix} \begin{Bmatrix} \varepsilon_{xx}^{(0)} \\ \varepsilon_{yy}^{(0)} \\ \varepsilon_{xy}^{(0)} \end{Bmatrix} + \begin{bmatrix} F_{11} & F_{12} & 0 \\ F_{12} & F_{22} & 0 \\ 0 & 0 & F_{66} \end{bmatrix} \begin{Bmatrix} \varepsilon_{xx}^{(1)} \\ \varepsilon_{yy}^{(1)} \\ \varepsilon_{xy}^{(1)} \end{Bmatrix} \\ + \begin{bmatrix} H_{11} & H_{12} & 0 \\ H_{12} & H_{22} & 0 \\ 0 & 0 & H_{66} \end{bmatrix} \begin{Bmatrix} \varepsilon_{xx}^{(3)} \\ \varepsilon_{yy}^{(3)} \\ \varepsilon_{xy}^{(3)} \end{Bmatrix}$$

$$\begin{Bmatrix} R_y \\ R_x \end{Bmatrix} = \begin{bmatrix} D_{44} & 0 \\ 0 & D_{55} \end{bmatrix} \begin{Bmatrix} \varepsilon_{yz}^{(0)} \\ \varepsilon_{xz}^{(0)} \end{Bmatrix} + \begin{bmatrix} F_{44} & 0 \\ 0 & F_{55} \end{bmatrix} \begin{Bmatrix} \varepsilon_{yz}^{(2)} \\ \varepsilon_{xz}^{(2)} \end{Bmatrix}$$

(7.2-27)